

EFFECT OF DIFFERENT CERAMIC REINFORCEMENTS ON MICROSTRUCTURE, MECHANICAL PROPERTIES AND TRIBOLOGICAL BEHAVIOUR OF THE Al6082 ALLOY PRODUCED BY STIR CASTING PROCESS

In the present study, the mechanical properties and high-temperature sliding wear behaviour of the Al6082-SiC-TiO₂ hybrid composite in different environmental conditions produced by the stir-casting process were investigated and distinguished with single-reinforced composites (Al6082-SiC and Al6082-TiO₂) and matrix alloy. The microstructure of composites exhibited a reasonably uniform scatter of particles in the aluminium matrix with good bonding between the matrix-particle interfaces. The hybrid composite's hardness and ultimate tensile strength showed higher hardness and tensile strength than matrix alloy and single-reinforced composites, whereas trends were reversed for the elongation. The impact test of the materials was conducted at different temperatures (room temperature, 0°C, -25°C, -50°C, and -75°C). The hybrid composite shows higher impact strength than the other materials, and impact strength decreases with temperature because ductility decreases with temperature. The fracture surfaces were examined to identify the fracture mechanism. The sliding wear test was conducted at different temperatures (room temperature, 100°C, 175°C, 250°C and 325°C) to distinguish the tribological behaviour of materials. The weight loss of the materials was increased with an increase in temperatures. The hybrid composite shows a lower weight loss than the other condition samples, irrespective of the temperatures. The wear surfaces were examined to predict the material removal mechanism.

Keywords: Al6082 alloy; Low-temperature impact properties; SiC particles; TiO₂ particles; Hybrid Composites; High-temperature wear properties

1. Introduction

In the present scenario, manufacturing lightweight automotive vehicles is in high demand on a global scale. As a result, continuing research is being carried out to identify materials with adequate strength-to-weight ratios, such as aluminium alloys. Aluminium alloys are extensively used in numerous engineering fields, such as automotive, aerospace, military, and marine, owing to their exceptional qualities like outstanding corrosion resistance, low weight, and excellent stiffness [1-2]. However, poor wear behaviour is a negative aspect of aluminium alloys, and this aspect is solved by distributing reinforcements (like SiC, Al₂O₃, SiO₂, B₄C, TiB₂, ZrO₂, Si₃N₄ etc.) in aluminium alloys, also referred to as aluminium matrix composites (AMCs) [3-9]. AMCs are widely used in automotive and aeronautical industries because of their superior strength-to-weight ratio, good corrosion resistance, high wear resistance, favourable mechanical properties, higher thermal diffusivity, low thermal expansion, and higher thermal conductivity [10-11]. Aluminium hybrid

matrix composites (AHMCs) are new-generation composites with more than one type of reinforcement with different shapes and sizes to gain better properties [12-16].

The literature reveals that the mechanical properties of AMCs were thoroughly investigated and well understood, but there needs to be more information available about these materials' toughness. However, these composites' toughness-impact strength has been studied in many research works at room temperature using various materials and processing parameters [17-22]. In contrast, very few studies on the behaviour of Al-SiC composites at different temperatures [23-24] and no studies were done on the Al-SiC-TiO₂ composites. For a temperature range of 25°C (room temperature) to -75°C, the Al-SiC-TiO₂ composites have significant applications in various industries; therefore impact behaviour of Al/SiC/TiO₂ composites needs to be investigated.

AMCs are a reliable alternative for aluminium alloys, cast iron, and steel to achieve energy savings, durability, and environmental-economic benefits. As a result, AMCs are employed

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in many industrial and automotive parts where wear and tear, weight savings, and seizure are major issues. These items include driveshafts, cylinder heads, connecting rods, pistons, and brake drums for the automobile industry. The mining and maritime sectors include impellers, valves, and turbine blades [25-32]. Usually, those components have experienced different kinds of surface defects like wear and tear. Thus, there is a need to study the wear characterisation of AMCs under different conditions. Several investigations have focused on the abrasive wear and sliding behaviour of AMCs [33-45,3,7]. In contrast, very few studies have focused on the dry sliding wear behaviour of AMCs at high temperatures.

Against this background, the present work examined the microstructure, mechanical characteristics and material removal mechanism on the high-temperature sliding wear behaviour of the Al-SiC-TiO₂ composites and their base alloys. Additionally, the fracture surface and wear surfaces were also examined through a scanning electron microscope (SEM).

2. Materials and methods

2.1. Matrix material and reinforcements

Al6082 was selected as a metal matrix material for the present investigation. The optical emission spectrometer (SPECTROMAXx LMF05, SPECTRO, made in Germany) was used for verifying the chemical ingredients of Al6082 alloy, as shown in TABLE 1. The SiC and TiO₂ particles, with a mean size of 10 μm and 5 μm, respectively, were taken for the reinforcement.

TABLE 1

Chemical ingredients of Al6082 alloy

Element	Mg	Al	Si	Mn	Ti	Fe	Cu	Zn	Zr
%wt	1.00	97.38	1.08	0.34	0.04	0.06	0.08	0.02	0.00

2.2. Experimentation

The experiments were designed to compare Al6082 alloys, composites (SiC/TiO₂ as reinforcements) and hybrid composites shown in TABLE 2. The investigation was carried out by melting the Al6082 alloy ingot in an induction furnace and heated up to 760°C. The super-heated molten metal was taken into a ladle,

TABLE 2

Design of Experiments

Exp No.	Condition
E-1	Al6082 alloy
E-2	Al6082+3%SiC composite
E-3	Al6082+3% TiO ₂ composite
E-4	Al6082+1.5%SiC+1.5%TiO ₂ hybrid composite

stirring induced into molten metal using the 4-axial mechanical stirrer blade speed of 700 rpm. The preheated ceramic powders were introduced into molten metal while stirring, and molten metal was degassed using argon gas. Uniform distribution of ceramic powders was done by stirring for 10 minutes. The molten metal slurry was poured into a preheated hardened mild steel mould; subsequent solidification was carried out.

2.3. Characterisation

2.3.1. Microstructural characterisation

Microstructural analysis of prepared samples was carried out using a scanning electron microscope (model-JEOL JSM-6480LV) outfitted with an Energy-Dispersive X-ray analyser (EDX). The samples were prepared with a dimension of 10 mm×10 mm. The samples were polished using different grades of emery paper (320, 600, 800, 1200, 1500, and 2000 grades), followed by velvet cloth polishing using diamond paste and ethanol. Keller's reagent was used as the etching reagent (1.5 ml HCl, 2.5 ml HNO₃, 1.0 ml HF, and 95 ml water).

2.3.2. XRD characterisation

An Ultima IV X-ray diffractometer with CuK alpha radiation and Nickel filter was used for phase analysis of different composites. 30 mA current intensity and a voltage of 40 kV, respectively, were the analysis parameters.

2.3.3. Mechanical properties

Tensile and impact samples were prepared using a wire-cutting machine (EDM). The tensile test specimens were prepared according to ASTM standards E8/E8M-04. Tensile tests were carried out by applying a universal hydraulic testing machine (Instron 8501) with a capacity of 125 kN, and the tests were conducted at a ram speed of 0.35 mm per min and an initial strain rate of $1.69 \times 10^{-4} \text{ s}^{-1}$.

The Impact tests were performed using instrumented Charpy tester (model-Zwick GmbH) with a low-temperature capacity of up to -196°C. The impact tests were performed at different temperatures of (room temperature, 0°C, -25°C, -50°C, and -75°C). Impact test carried samples with V-notch was prepared as per ASTM E23-12C of dimensions 55 mm×10 mm×10 mm and kept in liquid nitrogen for 10 minutes before the tests. The time gap between the taking away of samples from the liquid nitrogen to the tests was at least 5 seconds according to the standard of ASTM E23. After the tensile and impact test, the fracture surfaces of the materials were investigated using Scanning Electron Microscopy (Model-JEOL JSM-6480LV).

The hardness tests were performed using the Brinell hardness testing machine. The testing load was 250N applied

for 30 seconds, and the ball size of the indenter was 5 mm in diameter. A sample size of 20 mm × 20 mm was prepared for the hardness testing. An average of three indentations was taken for hardness value calculation in each sample.

2.3.4. Wear characterisation

The sliding wear tests of samples were conducted at different temperatures (RT, 100°C, 175°C, 250°C and 325°C) at loads 10 kN for a speed of 500 rpm. The sliding distance was fixed at 1200 m. A pin-on-disc sliding wear testing instrument (model: DUCOM-TR 20-LE) was used for the wear test, and a standard steel disc having a surface roughness of less than one micron was used. The wear samples (pins) size was 10 mm in diameter and 35 mm in length. The wear samples were dipped and appropriately cleaned before and after the test performance. For high-temperature wear tests, the pin-on-disc was surrounded by a hot chamber furnace to maintain a constant temperature. Three individual wear and friction tests were performed, and the mean value was taken. The rate of wear was calculated by the weight loss method. After the wear test, the wear surfaces of the materials were investigated using SEM (model-JEOL JSM-6480LV).

3. Results and discussion

3.1. Microstructural analysis

SEM images were taken for microstructural analysis, and EDX analysis was performed for elemental analysis of different materials (E-1, E-2, E-3, and E-4). Fig. 1(a) depicts the microstructure of E-1, and grain boundaries of α -Aluminum were encircled. Fig. 1(d) illustrates the EDX analysis of Al6082 alloy, confirming that all the major and minor elements present with appropriate weight %. Figs. 1(b and c) shows the morphology of reinforcement materials (SiC and TiO₂). The microstructure of composite samples was also examined using SEM. Fig. 2(a) represents the microstructure of E-2 and exhibits that the SiC particles (encircled) are distributed uniformly in the aluminium matrix. Fig. 2(b) shows the microstructure of E-3 and that the TiO₂ particles are distributed throughout the matrix. In contrast, Fig. 2(c) shows the microstructure of E-4 with a fine microstructure and clustering of particles. Fig. 2(d) shows the elemental analysis of hybrid composites through the EDX area mapping method, confirming the presence of SiC and TiO₂ particles and other elements with appropriate weight percentages throughout the matrix. The aluminium matrix can be seen as a dark region and reinforcements (SiC and TiO₂ particles) as grey and white

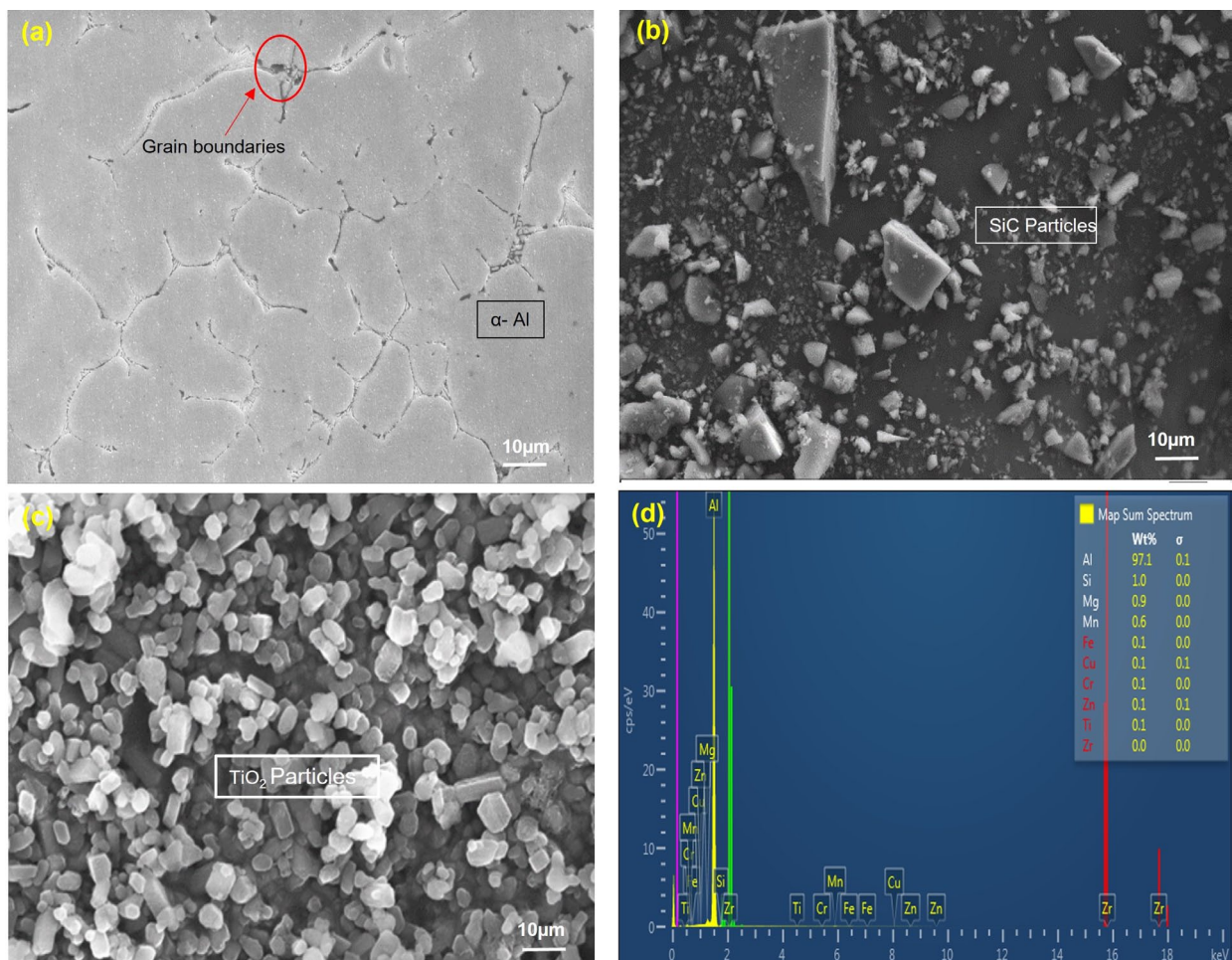


Fig. 1. (a) Microstructure of E-1, (b) Morphology of SiC particles, (c) Morphology of TiO₂ Particles, (d) EDX analysis of the E-1

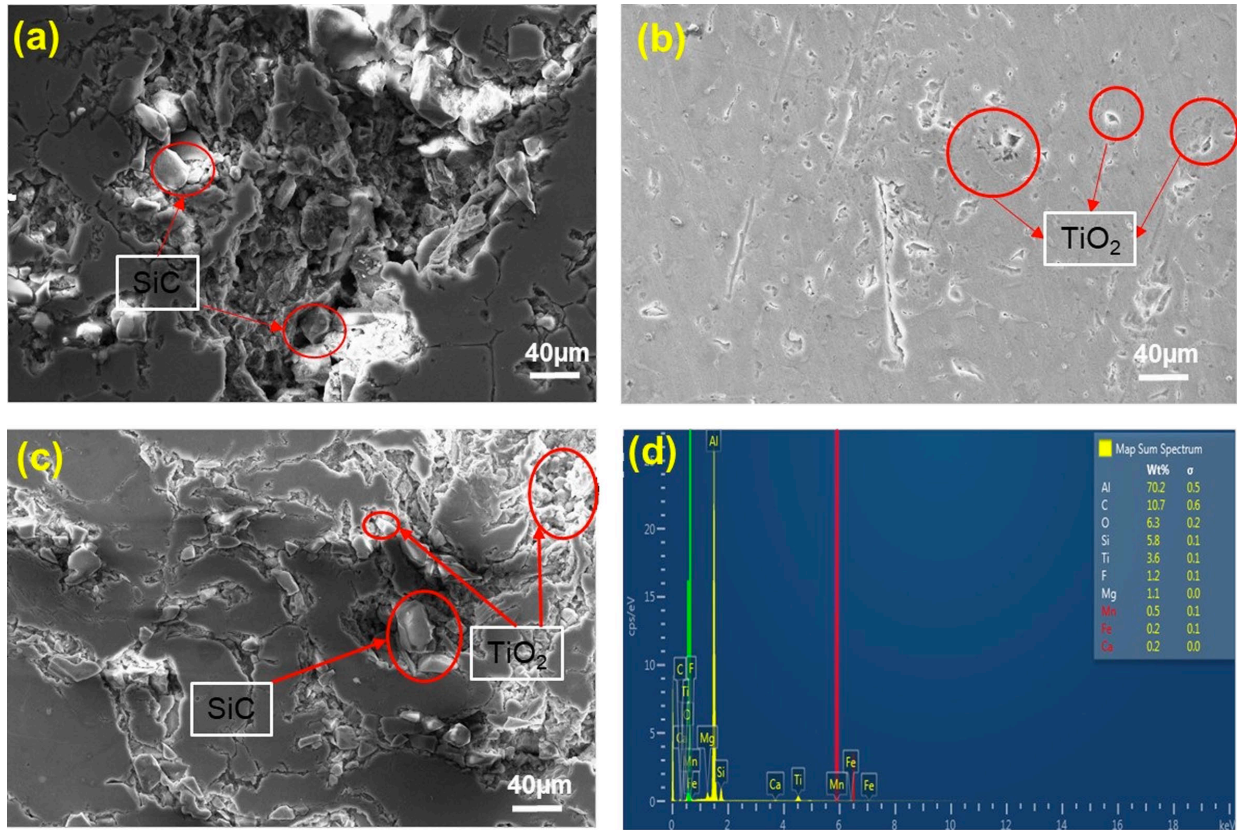


Fig. 2. Microstructure of (a) E-2, (b) E-3, (c) E-4, (d) EDX analysis of E-4

spots. The composites exhibited uniform distribution of reinforcements with minimal voids and discontinuities, which helps in an excellent interfacial bonding between metal matrix material and reinforcements.

3.2. Phase analysis

The phase analysis of samples for various materials (E-1, E-2, E-3, and E-4) was done by XRD technique, and the results are shown in Fig. 3. The phase analysis outcomes validate the occurrence of SiC and TiO₂ particles in the Al matrix. The sharp diffraction peaks of Al were at the 2 θ angles of 38.6°(111), 44.8°(200), 65.2°(220), and 78.3°(311), very weak peaks of SiC were at the 2 θ angles of 35.7°(111) and 60.2°(220) and TiO₂ peaks were at the 2 θ angle of 48.2°(004). The XRD results confirm that there were no other peaks or impurities present. A small portion of reinforcements was hard to identify in the XRD spectrum. In contrast, the intensity of these peaks enhanced with increasing the weight percentage of reinforcement [46].

3.3. Tensile test and hardness

Fig. 4 depicts the tensile strength and hardness of samples for different materials (E-1, E-2, E-3, and E-4). The experimental results show that E-4 samples exhibit more strength and hardness than the other samples. The hybrid composite's maximum

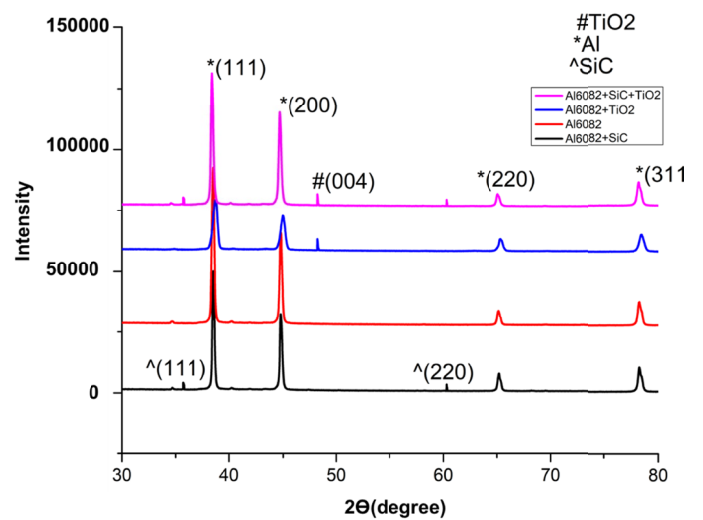


Fig. 3. XRD analysis of different conditions

hardness and strength were observed at 83 BHN and 205 MPa, respectively. The hardness and tensile strength of the hybrid composite were increased by 12.5%, 6.4%, and 22% and 12.6%, 6.5%, and 32.5% to Al6082-SiC composite, Al6082-TiO₂ composite, and matrix alloy, respectively. This occurred due to the addition of reinforcements (SiC and TiO₂) to the aluminium metal matrix, provided high resistance to plastic deformation, encouraged the increase in the high hardness and tensile strength of the hybrid composite, and increased the load-bearing ability of the hybrid composite. The reinforcement addition also increases

the strength and hardness of the single reinforced composites compared to the unreinforced aluminium matrix alloys. For apparent differences, the corresponding mechanical data are shown in TABLE 3.

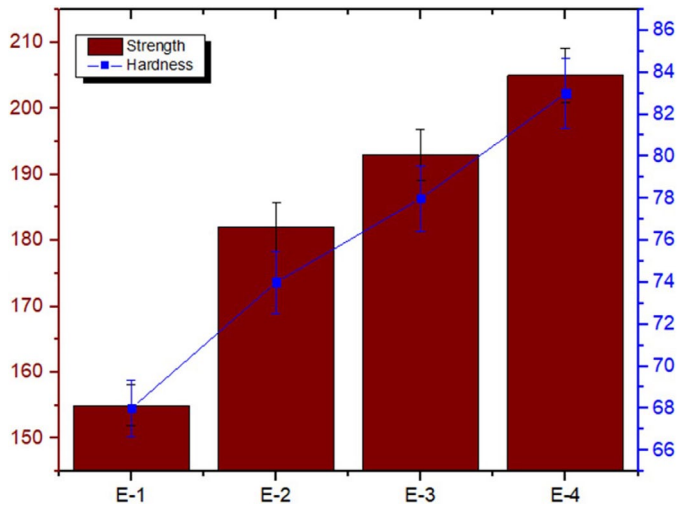


Fig. 4. Strength and hardness of the materials

It was observed that enhanced strength was due to the dispersion hardening effect, the load transfer mechanism, and or own strengthening mechanisms [47]. However, the ductility of reinforced samples was decreased due to the formation of voids and nucleation in the matrix around reinforcement materials [48]. The elongation of the hybrid composite was reduced by 7.5%, 2.5%, and 36.5% to Al6082-SiC composite, Al6082-TiO₂ composite and matrix alloy, respectively. One of the accepted strengthening mechanisms is the load transfer mechanism which occurs due to transmit of load (Eq. (1)) from matrix material to the solid particles of reinforcements through their interface [46].

$$\Delta\sigma_{load} = 0.5 f\sigma_m \tag{1}$$

Where f = fractional volume of reinforcements σ_m = yield strength of the matrix alloy in (MPa).

TABLE 3

Ultimate tensile strength (UTS) and hardness values of different samples

Sample	UTS (MPa)	%Elongation	Hardness (BHN)
E-1	155	8.02	67
E-2	182	5.21	74
E-3	193	5.5	78
E-4	205	5.1	83

3.4. Impact test analysis

The Charpy impact test was performed for the materials (E-1, E-2, E-3, and E-4) at different sub-zero temperatures (room temperature, 0°C, -25°C, -50°C and -75°C), and results are

shown in TABLE 4. Fig. 5 depicts that impact energy decreases at lower temperatures irrespective of the materials. It is because at lower temperatures brittleness nature of samples increases. In impact testing at all temperature conditions, impact energy increases with the addition of reinforcement SiC and TiO₂ particles to the matrix material (E-4), showing higher resistance to impact load when compared with other conditions (E-1, E-2, and E-3). The hybrid composite's highest impact energy was observed at room temperature. The maximum impact energy of hybrid composite (E-4) was 6.12 J which was increased by 11%, 10%, and 22.5% to Al6082-SiC composite, Al6082-TiO₂ composite, and matrix alloy, respectively, at room temperature. The decrease of impact strength with a reduction in temperature is an expected result as the ductility decreases with temperature [49].

TABLE 4

Impact energy in Joules at different temperatures

Condition	RT	0°C	-25°C	-50°C	-75°C
E-1	5.01	4.89	4.76	4.4	3.07
E-2	5.51	5.21	5.18	5.06	4.3
E-3	5.55	5.36	5.24	4.54	4.21
E-4	6.12	5.87	5.76	5.51	4.3

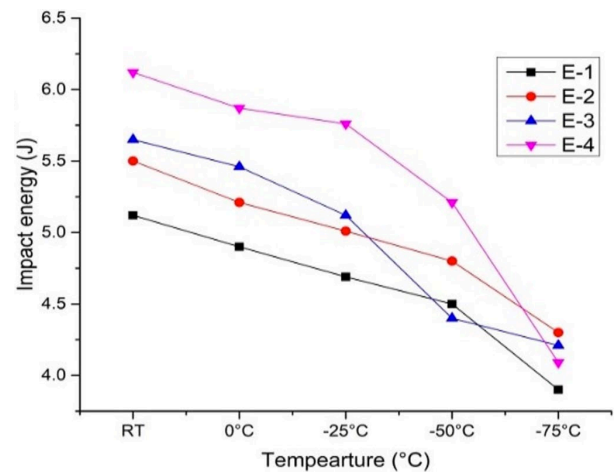


Fig. 5. Impact energy of the materials at different temperatures

3.5. Fractography analysis

3.5.1. Fracture surface of tensile samples

SEM images of fracture surfaces were taken for fractographic analyses, shown in Fig. 6. The fracture surface of the matrix alloy (E-1) was observed with dimples and cleavages; this confirms the ductile fracture. The fracture surfaces of E-2, E-3, and E-4 as shown in the Figs. 6(b-d), respectively. The changes in the behaviour of fracture surfaces of single-reinforced composites (E-2 and E-3) and hybrid composites (E-4) were observed. The fracture surfaces consist of dimple patterns and tear rigid, showing ductile and brittle fractures of mixed mode. The ductile fracture mechanism includes

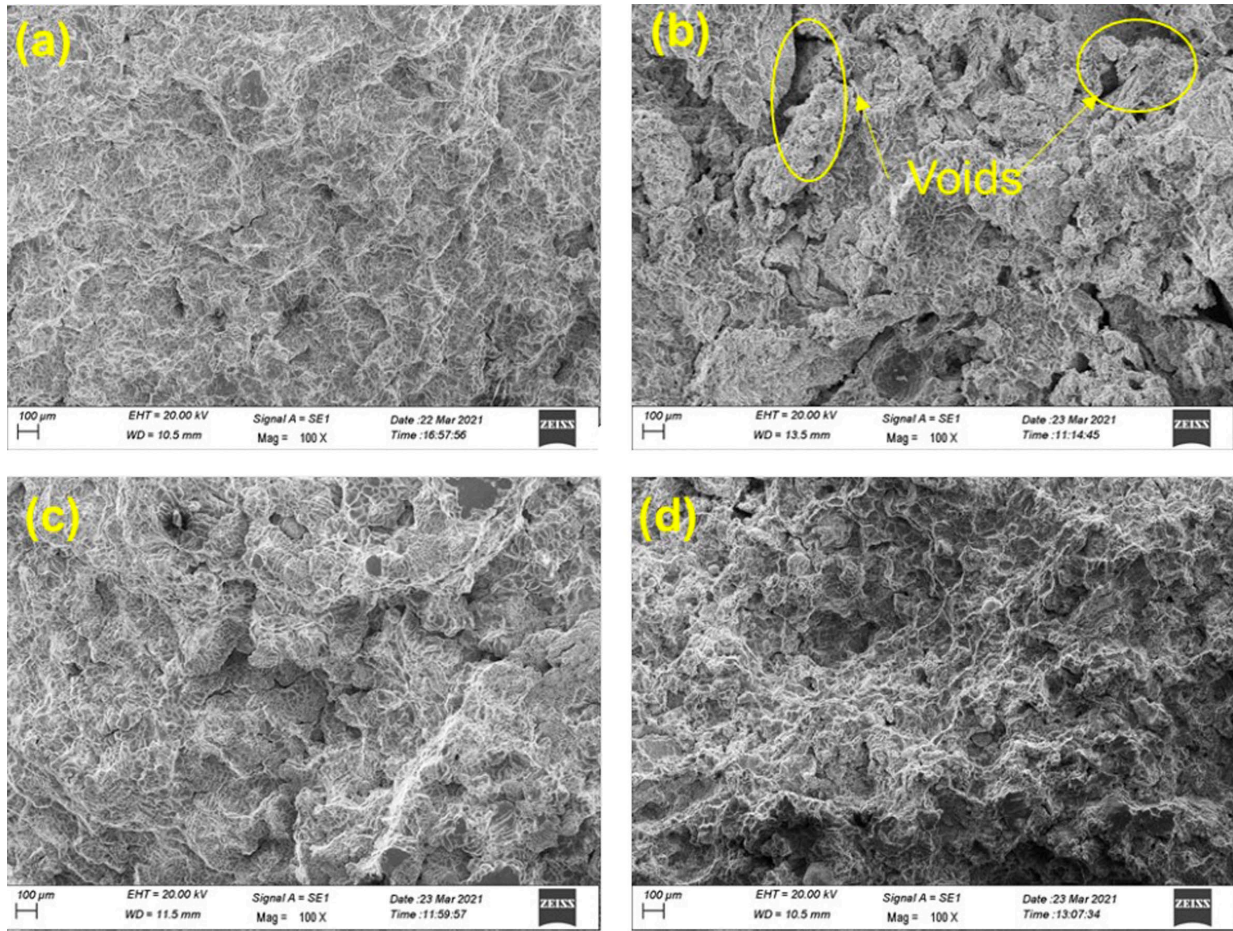


Fig. 6. The tensile fracture surfaces of (a) E-1, (b) E-2, (c) E-3, (d) E-4

nucleation, growth, and coalescence of voids in the matrix around reinforcement material. In contrast, brittle fracture in composites has features of particle pullout, crack growth, and crack propagation [49].

3.5.2. Fractography analysis of impact samples

The fracture surfaces of impact test samples for materials (E-1, E-2, E-3, and E-4) are shown in Fig. 7. The impact samples tested at room temperature depict a dimples structure. In contrast, samples at -75°C show the cleavage structure with microvoids and voids. The SEM images of fracture surfaces agree with the impact strengths obtained from testing.

3.6. Wear test analysis

Wear test was performed for materials (E-1, E-2, E-3 and E-4) at different high-temperature conditions (RT, 100°C , 175°C , 250°C and 325°C); after that wear results were analyzed. Figs. 8(a) and 8(b) show the variation in weight loss and the average coefficient of friction with different temperatures. Lower weight loss was observed in composite materials compared to Al6082 matrix alloys at different temperatures.

The hybrid composite exhibited excellent wear resistance to severe wear than the single reinforced composites at higher temperature regimes. The weight loss of all the materials was increased with an increase in temperature due to softening of materials. Weight loss of the E-2 composite was higher than the E-3 composite, up to 175°C and almost equal for both composites after 175°C . This is due to the precipitation (Mg_2Si) hardening of E-2 composites. The average friction coefficient of materials in different temperatures is shown in Fig. 8(b). At all temperatures, friction coefficients fluctuate about the mean level and decrease as the temperature increases. The friction coefficient fluctuations may be due to variations in contact between the sample and the disc [51]. The composites exhibited a lower frictional coefficient in comparison to the matrix alloy.

3.6.1. Analysis of wear surfaces

SEM was used to examine the wear surfaces after the wear test at room and high-temperature conditions. The morphological analysis of the wear surfaces shows that the wear debris obtained from the hybrid composite samples is fine and smooth, as shown in Fig. 9(d). The wear surfaces of matrix alloy (E-1) showed higher wear track marks and significantly wider grooves, as shown in Fig. 9(a). Over the sliding surface, a transfer layer of

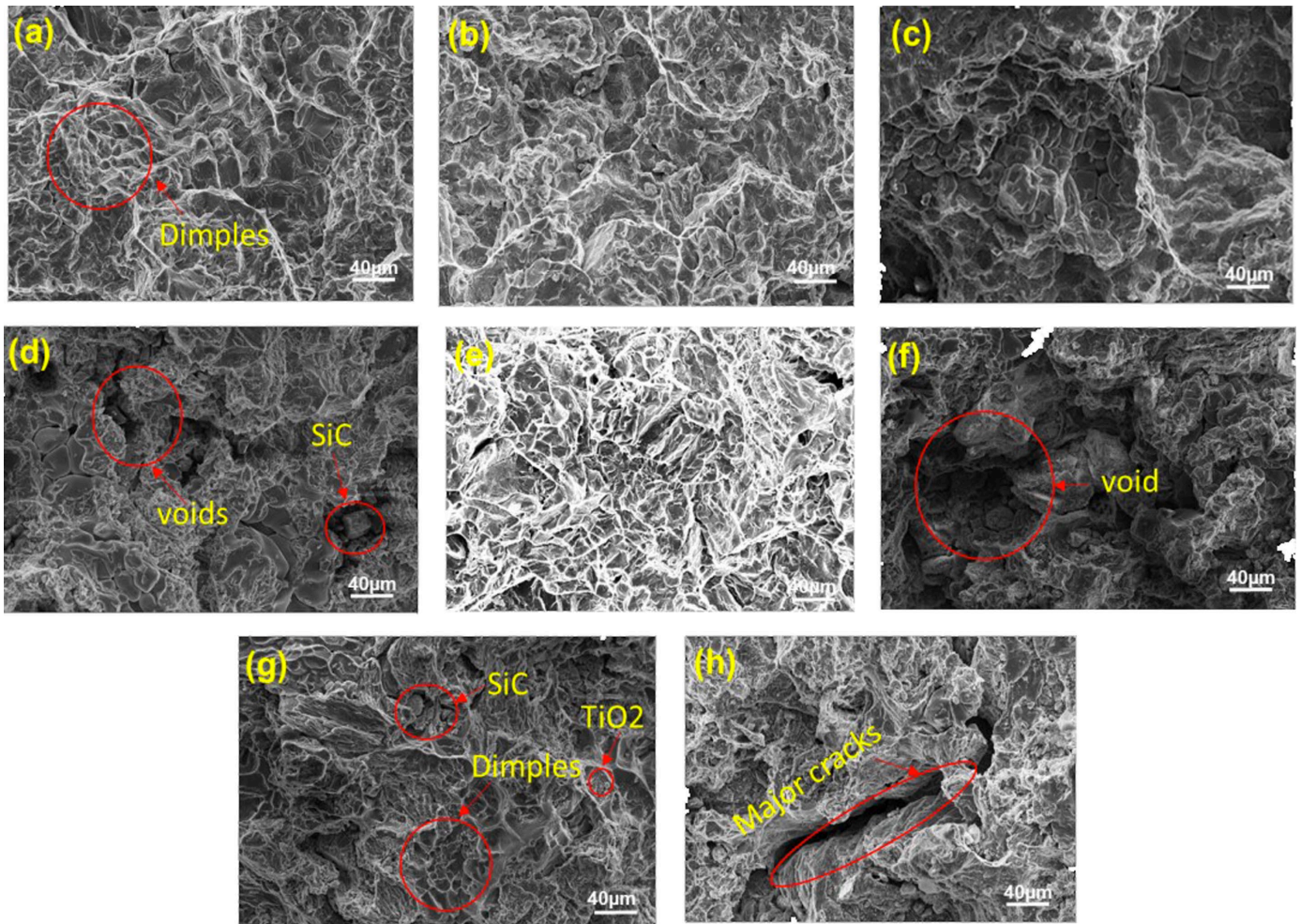


Fig. 7. Fractography of impact tests (a) matrix alloy at room temperature, (b) matrix alloy at -75°C , (c) E-2 at RT, (d) E-2 at -75°C , (e) E-3 at RT, (f) E-3 at -75°C , (g) E-4 at RT, (h) E-4 at -75°C

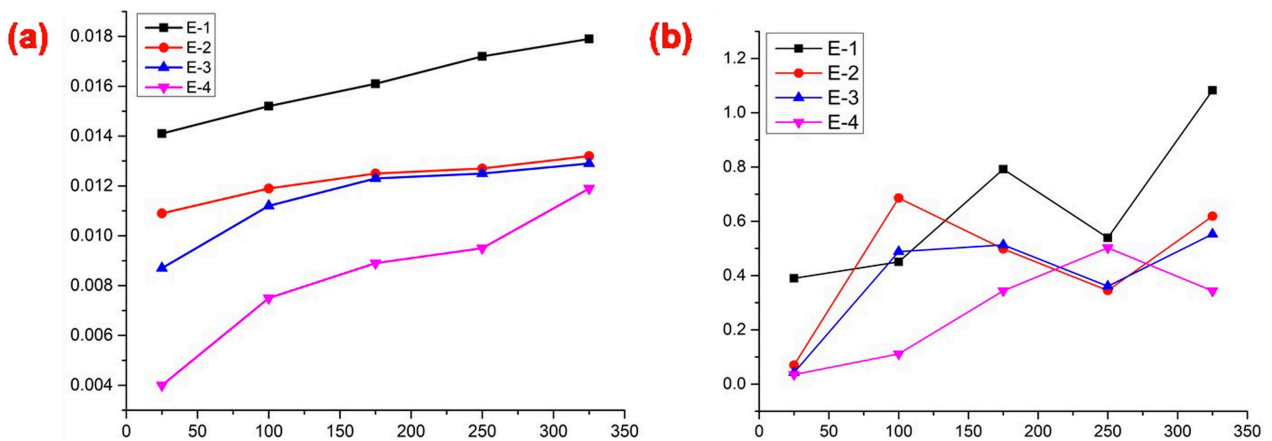


Fig. 8 (a) Weight loss of materials at different temperatures, (b) The friction coefficient of materials in different temperatures

compressed wear debris and wear tracks can be analysed. Before detaching, this layer reaches a specific thickness, which causes wear debris to form.

Comparatively, the cracks and wear debris in the Al6082-SiC composite (E-2) was less than that of the Al6082-TiO₂ composite (E-3), as shown in Figs. 9(b) and 9(c), due to the formation of precipitation at higher temperatures. Hybrid composite wear

surfaces showed shallower grooves and minimum craters when collated with Al6082-SiC composite, Al6082-TiO₂ composite, and matrix alloy. Due to shallower grooves, the weight loss in composites is lower. Another factor contributing to composites' lower weight loss is their increased hardness compared to matrix alloys, which results in a smaller real contact area and, thus, a lower rate of wear [39].

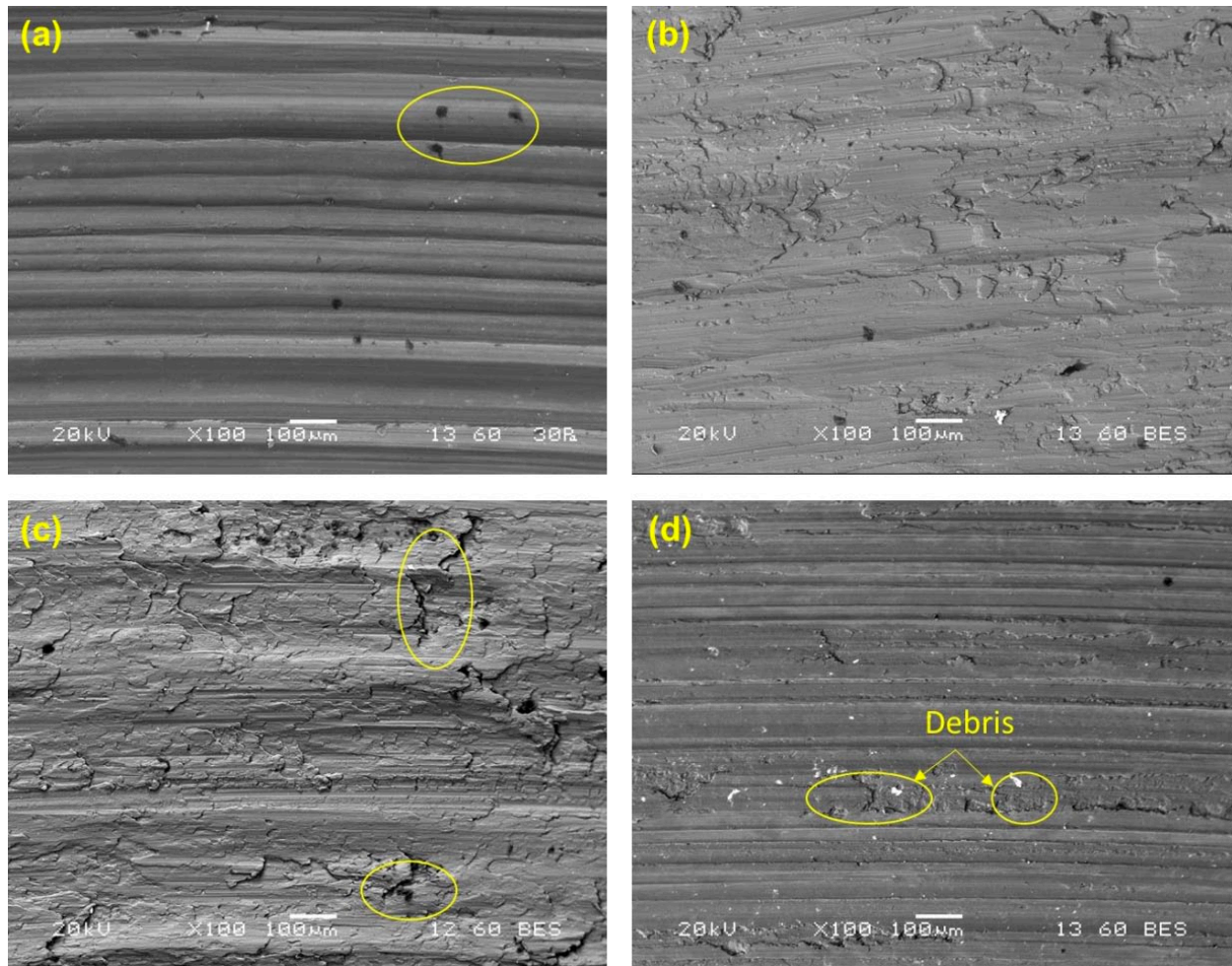


Fig. 9. The wear surfaces of (a) E-1, (b) E-2, (c) E-3, (d) E-4, at temperature of 250°C

4. Conclusions

The present study based on the microstructure, mechanical properties and high temperature sliding wear characterisation of Al6082-SiC-TiO₂ hybrid composite developed by stir casting process was investigated and distinguished to single reinforced composites (Al6082-SiC and Al6082-TiO₂) and matrix alloy. The outcomes of the researched study can be concluded as follows:

1. The microstructure of the single reinforced composites and hybrid composites revealed that the aluminium matrix consists of uniform distribution of reinforcement particles and minimal micro-voids and discontinuity.
2. The hybrid composites' hardness and ultimate tensile strength (UTS) showed higher hardness and tensile strength than matrix alloy and single-reinforced composites.
3. The impact energy decreased with decreasing temperatures, irrespective of the materials. The impact energy of hybrid composite was increased by 11%, 10%, and 22.5% to Al6082-SiC composite, Al6082-TiO₂ composite, and matrix alloy, respectively, at room temperature.
4. The fracture surfaces of both tensile and impact test samples exhibited cleavage fractures with minimum voids; it is evidence of a brittle fracture.

5. The hybrid composites have excellent wear resistance, irrespective of temperature, compared to the other materials.
6. The wear surfaces of hybrid composite showed shallower grooves compared to Al6082-SiC composite, Al6082-TiO₂ composite, and matrix alloy.

Hence, adding hard dispersoid particles in matrix materials increases the mechanical and tribological properties in different environmental temperatures. The hybrid composite shows superior properties to matrix alloy and single-reinforced composites.

Statements and declarations

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